

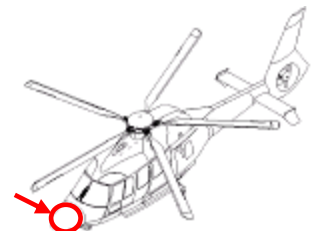
Civil versions: B, B1



# ALERT SERVICE BULLETIN

**SUBJECT: NAVIGATION - Attitude and Heading Reference System (AHRS)****Modification of the wiring of the FLIGHT/GROUND logic  
(connector 11 ALPHA)**

Corresponds to modifications 365A084754.00 and 365A084754.01

**For the attention of**

Revision No.	Date of issue
Revision 0	2017-07-19

**Summary:**

The purpose of this ALERT SERVICE BULLETIN is to correct the connections of the FLIGHT/GROUND logic of the primary reference system (Attitude and Heading Reference System AHRS), the FCDS (Flight and Control Display System) and the SSCVFDR (Combined Voice and Flight Data Recording System) by modifying the connection and, as per configuration, by modifying the wiring of connector 11 ALPHA.

**Compliance:**

Airbus Helicopters makes compliance with this ALERT SERVICE BULLETIN mandatory.

## **1. PLANNING INFORMATION**

### **1.A. EFFECTIVITY**

#### **1.A.1. Helicopters/installed equipment or parts**

Helicopters PRE MOD 0731B89 (SSCVFDR upgraded following standard ED112), and PRE MOD 365A084754.00,

OR

Helicopters POST MOD 0731B89 (SSCVFDR upgraded following standard ED112), and PRE MOD 365A084754.00 and 365A084754.01.

#### **NOTE**

*Refer to the Aircraft Individual Inspection Record (or RIC AMS) to identify the actual modification status of the aircraft.*

#### **1.A.2. Non-installed equipment or parts**

Not applicable.

### **1.B. ASSOCIATED REQUIREMENTS**

Not applicable.

### **1.C. REASON**

During a flight test for repair, a difference was found in the attitude indications provided by AHRS and displayed on the pilot and copilot PFD (Primary Flight Display) and the standby horizon during rapid turn changes of the helicopter.

This difference of indication is due to a wiring anomaly in the FLIGHT/GROUND logic from the management card of FLIGHT/GROUND states, as the AHRS have a different dynamic according to the FLIGHT or GROUND state of the helicopter.

The analysis of this incident showed that the AHRS were receiving the same FLIGHT/GROUND signal of the same position sensor of the landing gear whereas each AHRS must receive information from a different source to avoid displaying erroneous attitudes and vertical speed on both PFD. This absence of segregation in the acquisition of the FLIGHT/GROUND information also affects the FCDS (Flight Control Display System) system which includes FCDM (Flight Control Display Module), SMD45 pilot screens, and for helicopters POST MOD 0731B89, the recorder L3-COM FA5001-ED112.

To correct the existing common mode in the acquisition of the FLIGHT/GROUND information used by the AHRS, the FCDS system and the recorder L3-COM FA5001-ED112, Airbus Helicopters developed the modifications:

- 365A084754.00 which consists in re-allocating the electronic card outputs by modifying the connection to connector 11 ALPHA (helicopters PRE and POST MOD 0731B89),
- 365A084754.01 which consists in modifying the wiring of connector 11 ALPHA (helicopters POST MOD 0731B89).

**Airbus Helicopters makes compliance with this ALERT SERVICE BULLETIN mandatory.**

## **1.D. DESCRIPTION**

This ALERT SERVICE BULLETIN consists in:

Helicopters PRE MOD 0731B89: (compliance with modification 365A084754.00)

- modifying the connection to connector 11 APLHA,
- performing the associated tests.

Helicopters POST MOD 0731B89: (compliance with modifications 365A084754.00 and 365A084754.01)

- modifying the connection to connector 11 APLHA,
- modifying the wiring to connector 11 APLHA,
- performing the associated tests.

## **1.E. COMPLIANCE**

### **1.E.1. Compliance at H/C manufacturer level**

Helicopters/installed equipment or parts:

Airbus Helicopters makes compliance with instructions in paragraph 3. of this ALERT SERVICE BULLETIN mandatory before the delivery of the helicopter.

Non-installed equipment or parts:

Not applicable.

### **1.E.2. Compliance in service**

The work must be performed on the helicopter by the Operator.

Helicopters/installed equipment or parts:

Airbus Helicopters makes compliance with this ALERT SERVICE BULLETIN mandatory.

For the helicopters PRE MOD 0731B89, comply with [paragraph 3](#) (except for 3.B.2.b.), within 12 months as from receipt of this ALERT SERVICE BULLETIN, published at the date mentioned in page footer.

For the helicopters POST MOD 0731B89, comply with [paragraph 3](#) (except for 3.B.2.a.), within 12 months as from receipt of this ALERT SERVICE BULLETIN, published at the date mentioned in page footer.

Non-installed equipment or parts:

Not applicable.

**1.F. APPROVAL**Approval of modifications:

The information or instructions relating to modifications 365A084754.00 and 365A084754.01 which were approved on July 17, 2017 under the authority of EASA Design Organization Approval No. 21J.700 for helicopters of civil versions subject to an Airworthiness Certificate.

Approval of this document:

The technical information contained in this ALERT SERVICE BULLETIN Revision 0 was approved on July 18, 2017 under the authority of EASA Design Organization Approval No. 21J.700 for helicopters of civil versions subject to an Airworthiness Certificate.

**1.G. MANPOWER**

For compliance with this ALERT SERVICE BULLETIN, Airbus Helicopters recommends the following personnel qualifications:

Qualifications:      1 Electrical Technician.  
                              1 Avionics Technician.



Time for the operations is indicated for reference, for a standard configuration.

Time for the operations: 2 hours approximately for Electrical Technician for compliance with modification 365A084754.00.  
2 hours approximately for Electrical Technician for compliance with modification 365A084754.01.  
3 hours approximately for Avionics Technician (with jacking of the helicopter).



The estimated helicopter downtime is approximately 1 day.

**1.H. WEIGHT AND BALANCE**

Not applicable.

**1.I. POWER CONSUMPTION**

Not applicable.

**1.J. SOFTWARE UPGRADES/UPDATES**

Not applicable.

## 1 K. REFERENCES

The following documents are necessary for compliance with this ALERT SERVICE BULLETIN:

- AMM: 07.10.00.581: Lifting the Helicopter on Jacks.
- AMM: 24.00.00.911: General Safety Instructions - Electrical Power.
- AMM: 24.00.00.481: Power Supply - Electrical Power Systems.
- AMM: 34.00.00.911: General Safety Instructions - Navigation System.
- AMM: 34.23.00.721: Functional Tests - Primary Reference System.
  
- MTC: 20.07.01.201: Handling of helicopters in a hangar and in a prepared area.
- MTC: 20.07.02.201: Helicopter parked in a repair shop.
- MTC: 20.07.03.408: Appearance checks on an aircraft after an inspection or repair.
- MTC: 20.80.20.101: Contact Insertion and Extraction Method and Tools.
- MTC: 20.80.20.441: Installation of electrical cable bundles and optical fibres.

## 1.L. OTHER AFFECTED PUBLICATIONS



The modifications will be incorporated in:

- Wiring Diagram Manual (WDM),
- Special Index Modification (SIM).

These documents will be updated during a future revision.

## 1.M. PART INTERCHANGEABILITY OR MIXABILITY

Not applicable.

**2. EQUIPMENT OR PARTS INFORMATION****2.A. EQUIPMENT OR PARTS: PRICE - AVAILABILITY - PROCUREMENT**

For any information concerning modification kits and/or components or for assistance, contact the Sales and Customer Relations Department.

Airbus Helicopters  
Etablissement de Marignane  
Direction Ventes et Relations Client  
13725 MARIGNANE CEDEX  
FRANCE

**NOTE 1**

*On the purchase order, please specify the mode of transport, the destination and the serial numbers of the helicopters to be modified.*

**NOTE 2**

*For ALERT SERVICE BULLETINS, order by:  
Telex: HELICOP 410 969F  
Fax: +33 (0)4.42.85.99.96.*

**2.B. LOGISTIC INFORMATION**

Not applicable.

**2.C. EQUIPMENT OR PARTS REQUIRED PER HELICOPTER/COMPONENT****Components to be ordered for one helicopter POST MOD 0731B89**

Designation	Qty	New P/N	Item	Former P/N	Instruction
Extension	2	E0541-10	1	/	/
Cap	1	E0737B0816	2	/	/

**NOTE 3**

*The components listed in the table above concern the modification of the wiring of connector 11ALPHA (modification 365A084754.01). These components are not necessary for helicopters PRE MOD 0731B89.*

**Equipment or parts to be ordered separately for one helicopter POST MOD 0731B89:**

Designation	Qty	New P/N	Item	Former P/N	Instruction
Wire, electrical	2 m	EN2267-010A002S	3	/	/

**NOTE 4**

*The equipment or parts listed in the table above concern the modification of the wiring of connector 11ALPHA (modification 365A084754.01). The electrical wire (3) is not necessary for helicopters PRE MOD 0731B89.*

**Consumables to be ordered separately:**

As per Work Cards and Tasks mentioned in this ALERT SERVICE BULLETIN.

The consumables can be ordered separately from the KLX AEROSPACE SOLUTIONS company.

Website: <https://www.klxaerospace.com/klxaero/>

Telephone: +1.305.925.2600

AOG: +1.305.471.8888

**2.D. EQUIPMENT OR PARTS TO BE RETURNED**

Not applicable.



### 3. ACCOMPLISHMENT INSTRUCTIONS

#### 3.A. GENERAL

- Read and comply with instructions for handling helicopters in a hangar as per MTC Work Card 20.07.01.201.
- Read and comply with instructions for helicopters parked in a repair shop as per MTC Work Card 20.07.02.201.
- Read and comply with instructions on contact insertion and extraction method and tools as per MTC Work Card 20.80.20.101.
- Read and comply with instructions for installation of electrical cable bundles as per MTC Work Card 20.80.20.441.

#### 3.B. WORK STEPS



##### **CAUTION**

**BEFORE STARTING WORK ON THE ELECTRICAL SYSTEMS, READ TASK 24-00-00-911.**



##### **CAUTION**

**BEFORE STARTING WORK ON THE NAVIGATION INSTALLATIONS, READ TASK 34-00-00-911.**

##### 3.B.1. Preliminary steps

- Install suitable access means.
- Remove and/or open all cowlings, panels, doors and equipment as required for access to work zones.

##### 3.B.2. Procedure (Figure 1)

###### 3.B.2.a. Helicopters PRE MOD 0731B89

- Perform the modification of the connection to connector 11 ALPHA as per [Figure 1](#).

###### 3.B.2.b. Helicopters POST MOD 0731B89

- Perform the modification of the connection to connector 11 ALPHA as per [Figure 1](#).
- Perform the modification of the wiring to connector 11 ALPHA as per [Figure 2](#).



**3.B.3. Final steps**

- Carry out appearance checks on an aircraft after an inspection or repair as per MTC Work Card 20.07.03.408.
- Install and/or close all cowlings, panels, doors and all equipment removed and/or opened during preliminary steps ([paragraph 3.B.1.](#)).
- Remove the access means.

**3.B.4. Tests**

- Perform the "tests following modification of FLIGHT/GROUND wiring on the rear landing gear struts" as per [Appendix 4.A.](#)
- Perform the operating tests of the primary reference system as per AMM Task 34.23.00.721.
- Reset helicopter into flight condition.

**3.C. COMPLIANCE CONFIRMATION**Compliance with this document:

Record compliance with this ALERT SERVICE BULLETIN with the Revision number in the aircraft documents.

Tracking of modifications in the documentation:Helicopters PRE MOD 0731B89:

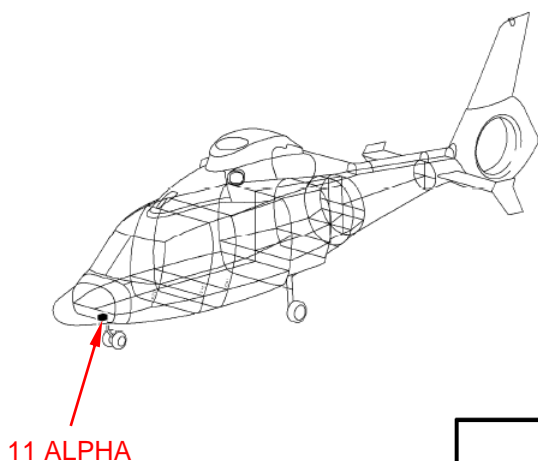
Record compliance with modification 365A084754.00 in the aircraft documents.

Helicopters POST MOD 0731B89:

Record compliance with modifications 365A084754.00 and 365A084754.01 in the aircraft documents.

**3.D. OPERATING AND MAINTENANCE INSTRUCTIONS**

Not applicable.

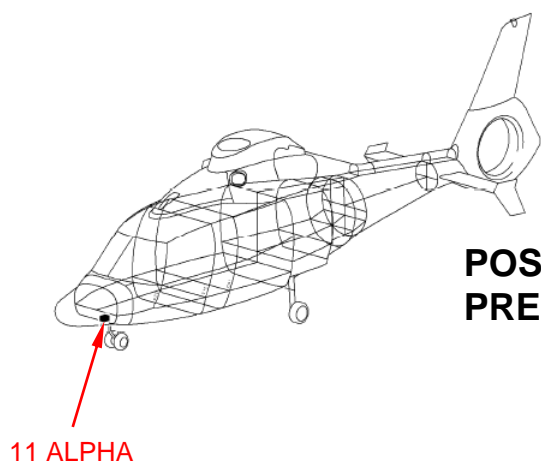

**PRE MOD 365A084754.00**

		11ALP2-CP1	
VOL 1	1	1UW22E	
	2	1UW25NE	
	5	1FF4E	
VOL 5	6	1FF45NE	
	7	1UW28E	
VOL 7	8	1UW29E	
		11ALP2-DP1	
VOL 2	1	2UW22E	
	2	2UW25NE	
VOL 6	5	2FF4E	
	6	2FF45NE	

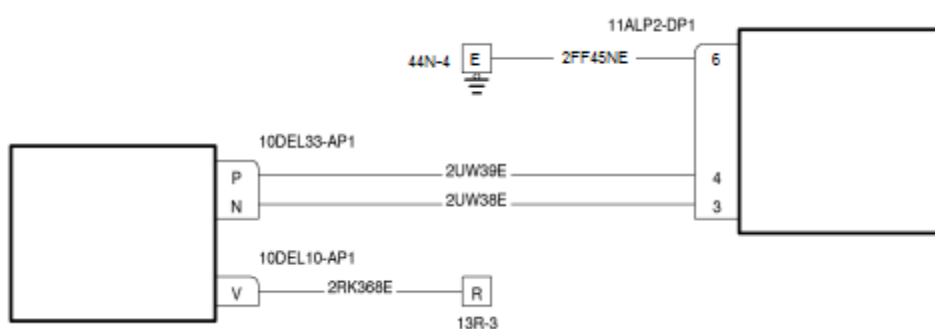
**POST MOD 365A084754.00**

		11ALP2-CP1	
VOL 5	5	1UW22E	
	6	1UW25NE	
VOL 7	7	1FF4E	
	8	1FF45NE	
VOL 1	1	1UW28E	
	2	1UW29E	
		11ALP2-DP1	
VOL 6	5	2UW22E	
	6	2UW25NE	
VOL 2	1	2FF4E	
	2	2FF45NE	

Figure 1



**POST MOD 0731B89  
PRE MOD 365A084754.01**



**POST MOD 0731B89  
POST MOD 365A084754.01**

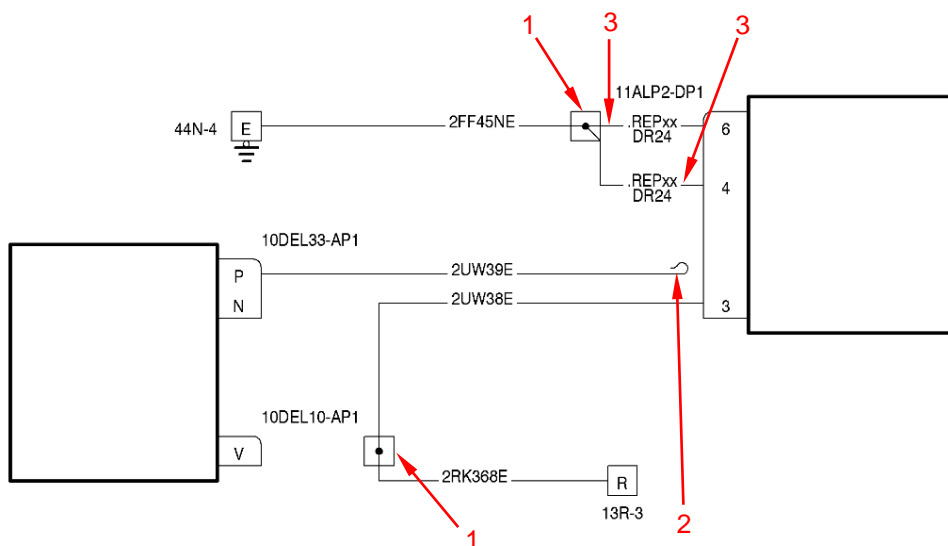


Figure 2

## 4. APPENDIX

### 4.A. TESTS FOLLOWING MODIFICATION OF THE FLIGHT/GROUND WIRING ON THE REAR LANDING GEAR STRUTS

#### 4.A.1. Preliminary steps

- Energize the helicopter using the electrical power bench as per Sub-Task 24-00-00-481-002 of Task 24-00-00-481.

#### 4.A.2. Procedure



#### CAUTION

**DO NOT ACTIVATE THE CVFDR MAINTENANCE COMMAND.**



#### CAUTION

**FOR CHECKING THE CVFDR OPERATION, IT IS NECESSARY TO RESPECT THE RIPS DISCHARGE TIME OF 10 MINUTES.**

- Jack up helicopter as per Sub-Task 07-10-00-581-001 of Task 07-10-00-581.
- Disconnect electrical connectors "9G" (RH landing gear) and "10G" (LH landing gear).

#### 4.A.2.a. LH landing gear to FLIGHT position / RH landing gear to FLIGHT position

- On LH landing gear, make a shunt between "D" and "E" (fixed connector) in order to simulate the FLIGHT position.
- On RH landing gear, make a shunt between "D" and "E" (fixed connector) in order to simulate the FLIGHT position.
- Energize the helicopter electrical system.

##### Acquisition of FCDM information on SMD45

- Turn off then on the 4 screens.
- Make sure that the cockpit symbols are displayed ("T" must not be displayed).

##### Acquisition of AHRS information

- On the AHRS control box, set "AHRS 1" to "OFF" then to "ON".
- Make sure that AHRS 1 switches to alignment then displays the information after approx. 1 minute and 5 seconds on both copilot screens.
- On the AHRS control box, set "AHRS 2" to "OFF" then to "ON".
- Make sure that AHRS 2 switches to alignment then displays the information after approx. 1 minute and 5 seconds on both pilot screens.

##### Checking AFCS logics

- Press "BC" button or "HOV" button more than 3 seconds.
- Briefly press "APP".
- Make sure that the "Flight FT" wording is displayed on the PFD.

Validation of the START/STOP logic: CVFDR L3 COM - ED112 (helicopters POST MOD 0731B89)

- Make sure that the CVFDR operates: the lights "**CVR**", "**FDR**" and "**HUMS**" must go off.
- Make sure that if one or the other of the 2 CVFDR circuit breakers is de-activated, the CVFDR remains active.

#### **4.A.2.b. LH landing gear to GROUND position / RH landing gear to FLIGHT position**

- De-energize the helicopter electrical system.
- On LH landing gear, remove the shunt between "**D**" and "**E**" (fixed connector) in order to simulate the GROUND position.
- Energize the helicopter electrical system.

Acquisition of FCDM information on SMD45

- Turn off then on the 4 screens.
- Make sure that the cockpit symbols are displayed ("**T**" must not be displayed).

Checking AFCS logics

- Press "**BC**" button or "**HOV**" button more than 3 seconds.
- Briefly press "**APP**".
- Make sure that the "**GROUND TT**" wording is displayed on the PFD.

Validation of the START/STOP logic: CVFDR L3 COM - ED112 (helicopters POST MOD 0731B89)

- Make sure that the CVFDR operates: the lights "**CVR**", "**FDR**" and "**HUMS**" must go off.
- Make sure that if one or the other of the 2 CVFDR circuit breakers is de-activated, the CVFDR remains active.

#### **4.A.2.c. LH landing gear to FLIGHT position / RH landing gear to GROUND position**

- De-energize the helicopter electrical system.
- On LH landing gear, make a shunt between "**D**" and "**E**" (fixed connector) in order to simulate the FLIGHT position.
- On RH landing gear, make a shunt between "**D**" and "**F**" (fixed connector) in order to simulate the GROUND position.
- Energize the helicopter electrical system.

Acquisition of FCDM information on SMD45

- Turn off then on the 4 screens.
- Make sure that the cockpit symbols are displayed ("**T**" must not be displayed).

Checking AFCS logics

- Press "**BC**" button or "**HOV**" button more than 3 seconds.
- Briefly press "**APP**".
- Make sure that the "**GROUND FF**" wording is displayed on the PFD.

Validation of the START/STOP logic: CVFDR L3 COM - ED112 (helicopters POST MOD 0731B89)

- Make sure that the CVFDR operates: the lights "**CVR**", "**FDR**" and "**HUMS**" must go off.
- Make sure that if one or the other of the 2 circuit breakers of the CVFDR is de-activated, the CVFDR remains active.

**4.A.2.d. LH landing gear to GROUND position / RH landing gear to GROUND position****NOTE**

*Provide access to the MGB for this test phase in order to act on the lug "16E".*

- De-energize the helicopter electrical system.
- On LH landing gear, remove the shunt between "D" and "E" (fixed connector).
- On RH landing gear, remove the shunt between "D" and "F" (fixed connector).
- Connect electrical connectors "9G" (RH landing gear) and "10G" (LH landing gear).
- Jack down helicopter as per Sub-Task 07-10-00-581-001 of Task 07-10-00-581.
- Energize the helicopter electrical system.

**Acquisition of FCDM information on SMD45**

- Turn off then on the 4 screens.
- Make sure that a "T" is displayed on the 4 screens for some seconds and then that the cockpit symbols are displayed (initialization phase longer than in flight).

**Acquisition of AHRS information**

- On the AHRS control box, set "AHRS 1" to "OFF" then to "ON".
- Make sure that AHRS 1 switches to alignment then displays the information after approx. 30 seconds on both copilot screens.
- On the AHRS control box, set "AHRS 2" to "OFF" then to "ON".
- Make sure that AHRS 2 switches to alignment then displays the information after approx. 30 seconds on both pilot screens.

**Checking AFCS logics**

- Press "BC" button or "HOV" button more than 3 seconds.
- Briefly press "APP".
- Make sure that the "GROUND TF" wording is displayed on the PFD.

**Validation of the START/STOP logic: CVFDR L3 COM - ED112 (helicopters POST MOD 0731B89)**

- On the MGB, disconnect the lug "16E" in order to simulate the MGB oil pressure.
- Wait for RIPS discharge (10 minutes).
- Make sure that the CVFDR operates: the lights "CVR", "FDR" and "HUMS" must go off.
- Make sure that if one or the other of the 2 CVFDR circuit breakers is de-activated, the CVFDR remains active.
- On the MGB, connect the lug "16E".
- Make sure that the CVFDR does not operate: the lights "CVR" and "FDR" are on.

**4.A.3. Final steps**

- De-energize the helicopter electrical system as per Sub-Task 24-00-00-481-002 of Task 24-00-00-481.